

SOLUTIONS MANUAL

Second Edition

ADVANCED MECHANICS OF MATERIALS

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Contents

List of Final Answers	L-1
Chapter 1	1
Chapter 2	9
Chapter 3	20
Chapter 4	28
Chapter 5	55
Chapter 6	68
Chapter 7	79
Chapter 8	95
Chapter 9	107
Chapter 10	123
Chapter 11	136
Chapter 12	151
Chapter 13	163
Chapter 14	180

List of Final Answers

For details, and where proofs are required, see following worked-out solutions

- 1.4-1 [Proof required]
 1.4-2 $P = 2kv(a - L)/a$ where $a^2 = L^2 + v^2$
 1.4-3 $p = (\alpha_a - \alpha_s)TtE_aE_s/[R(E_a - E_s)]$
 1.5-1 [Proof required]
 1.5-2 $\tau_{zx} = (P/A) - (Pd/J)y$, $\tau_{yz} = (Pd/J)x$, where $J = I_x + I_y$
 1.6-1 Consider equilibrium; note that $M = 0$ at inflection point
 1.6-2 [Proof required]
 1.6-3(a,b) [Proof required]
 (c) $\epsilon = ky$ if cross sections warp identically
 1.6-4 Surfaces: $\sigma_t = 3M(1 + \sqrt{R})/bh^2\sqrt{R}$, $\sigma_c = -\sqrt{R}\sigma_t$, where $R = E_c/E_t$
 1.6-5 $M_{\max} = F^2/2q$
 1.6-6(a) $F_a = 2EI/\rho L$ (b) $s = L - 2EI/\rho F_b$
 1.6-7 $h_L/h_O = 3$, stress ratio = 9/8
 1.7-1 $\sigma_A = -\tau L/c$, $\sigma_B = 2\tau L/c$, $v_C = -\tau L^3/2Ec^2$
 1.7-2 $v_C = -5hPL^2/48EI$
 1.7-3 $v_C = \alpha L^2\Delta T/2c$
 1.7-4 Doubtful, unless P is very small. Links become inclined.
 1.7-5 $u_O = L - \rho \sin \theta$, $v_O = \rho(1 - \cos \theta)$; where $\theta = L/\rho$, $\rho = EI/M_O$
 1.7-6 $a/b = 2.00$
 1.7-7 $a/b = 0.732$
 1.7-8(a) $y = Fx^3/3EI$ (b) $y = Fx^2(L - x)^2/3EIL$
 1.7-9 $R_x = R + P(x^4 - 2Lx^3 + L^3x)/12EIL$
 1.8-1(a) $u_D = 0$, $v_D = 4QL/\sqrt{3}AE$
 (b) $u_D = PL/2.30AE$, $v_D = 0$
 (c) $u_D = 1.188\alpha L\Delta T$, $v_D = 0$
 1.8-2 $T/\theta = 9GJ/20L$
 1.8-3 $0.0169PL^3/EI$ at load, $0.0039PL^3/EI$ at point opposite
 1.8-4 Angle = $TL^3/8EI[3R(R + L) + L^2]$
 1.8-5(a) $v = 5qL^4/384EI$ (b) $v = M_LL^2/16EI$, $\theta_L = M_LL/3EI$
 (c) $v = PL^3/192EI$ (d) $v = qL^4/768EI$
 1.8-6 Consider equilibrium. Match strain & curvature at interface.
 1.8-7 $\sigma = 3Et(D + t)/L^2$
 1.8-8 $L^4 = 72EID/q$
 1.8-9 $\Delta T = \theta h/\alpha L$
 1.9-1(a) $P = \sigma_Y AL/b$ (b) $P_{fp} = 2A\sigma_Y$
 (c) $\sigma_{res} = \sigma_Y(L - 2b)/L = \sigma_Y(2a - L)/L$
 1.9-2(a) $\sigma_1 = \sigma_Y$, $\sigma_2 = \sigma_Y/2$
 (b) $(\sigma_1)_{res} = -\sigma_Y/2$, $(\sigma_2)_{res} = -\sigma_Y/4$, $u_{res} = \sigma_Y L/4E$

- 1.9-3 $P_Y = 2.30\sigma_Y A$ at $u_D = \sigma_Y L/E$
 $P_{fp} = 2.73\sigma_Y A$ at $u_D = 4\sigma_Y L/3E$
- 2.3-1 (a) Pure shear
 (b) Hydrostatic in a plane, or uniaxial stress
 (c) Fully hydrostatic (3D)
- 2.3-2 (a) $\sigma_1 = 82.1$, $\sigma_2 = 0$, $\sigma_3 = -52.1$ (in MPa)
 (b) $\sigma_1 = 127.6$, $\sigma_2 = -12.9$, $\sigma_3 = -195$ (in MPa)
 (c) $\sigma_1 = 114$, $\sigma_2 = 68.4$, $\sigma_3 = -163$ (in MPa)
 (d) $\sigma_1 = 297$, $\sigma_2 = 99.6$, $\sigma_3 = -177$ (in MPa)
 (e) $\sigma_1 = 22.4$, $\sigma_2 = 0$, $\sigma_3 = -22.4$ (in MPa)
 (f) $\sigma_1 = 74.6$, $\sigma_2 = -19.4$, $\sigma_3 = -55.2$ (in MPa)
 (g) $\sigma_1 = 200$, $\sigma_2 = -100$, $\sigma_3 = -100$ (in MPa)
- 2.3-3 $\begin{Bmatrix} \ell_1 & m_1 & n_1 & \ell_2 & m_2 & n_2 & n_3 = n_1 \times n_2 \end{Bmatrix}$
- 2.3-4 (a) 0 0.973 0.230 0 -0.230 0.973
 (b) 0.080 0.792 0.605 0.787 -0.423 0.449
 (c) -0.309 0.925 0.223 0.911 0.219 0.353
 (d) 0.806 -0.582 0.111 0.427 0.700 0.573
 (e) 0.632 0.707 0.316 -0.447 0 0.894
 (f) 0.651 0.502 0.570 -0.083 0.793 -0.604
 (g) 0.577 0.577 0.577 [any normal to n_1]
- 2.3-5 $\sigma_1 = \sigma_2 = 0$, $\sigma_3 = -80$ MPa
- 2.4-1 (a,b) [Proof required]
- 2.5-1 (a,b) [Proof required] (c) $B = E/3(1 - 2\nu)$
 (d) [Proof required]
- 2.5-2 $\epsilon_1 = 0.000457$, $\epsilon_2 = -0.000066$, $\epsilon_3 = -0.000303$
- 2.5-3 $A = \text{tr}_i/(1 - \nu)$
- 2.5-4 $\Delta T = 347^\circ\text{C}$
- 2.5-5 [Set of equations required]
- 2.5-6 $\tan \theta = \sqrt{\nu}$, $\sigma_x = E\epsilon_s/(1 - \nu)$
- 2.5-7 $p = 2Et(r - a)/[(1 - \nu)r^2]$, maximum at $r = 2a$
- 2.6-1 Case 1 (a) $\sigma_1 = 30$ MPa, $\sigma_2 = 20$ MPa, $\sigma_3 = -20$ MPa
 (b) $n_1 = 1$, $\ell_2 = m_2 = -\ell_3 = m_3 = 0.707$
 (c) $\tau_{\text{oct}} = 21.6$ MPa, $\tau_{\text{max}} = 25$ MPa
 (d) $\tau_e = 45.8$ MPa
 (e) $U_{\text{od}} = 350/G \text{ N}\cdot\text{mm}/\text{mm}^3$ (G in MPa)
- Case 2 (a) $\sigma_1 = 35.1$ MPa, $\sigma_2 = 7.1$ MPa, $\sigma_3 = -27.2$ MPa
 (b) $\ell_1 = 0.636$, $m_1 = 0.384$, $n_1 = 0.669$
 $\ell_2 = 0.240$, $m_2 = 0.725$, $n_2 = -0.645$
 (c) $\tau_{\text{oct}} = 25.5$ MPa, $\tau_{\text{max}} = 31.2$ MPa
 (d) $\sigma_e = 54.1$ MPa
 (e) $U_{\text{od}} = 487/G \text{ N}\cdot\text{mm}/\text{mm}^3$ (G in MPa)
- 2.6-2 (a) (b) (c) (d), s_x (e), s_1 (f)
 (a) 67.1 55.2 117 -10 72.1 2285/G
 (b) 161 132 280 -53.3 154 13,070/G
 (c) 138 121 257 48.3 107 11,000/G

- | | | | | | | |
|-----|------|------|------|-----|------|----------|
| (d) | 237 | 194 | 412 | 107 | 224 | 28,300/G |
| (e) | 22.4 | 18.3 | 38.7 | 0 | 22.4 | 250/G |
| (f) | 64.9 | 54.7 | 116 | 0 | 74.6 | 2250/G |
| (g) | 150 | 141 | 300 | 0 | 200 | 15,000/G |
- 2.7-1 $K_t = 2.0$ for small load, $K_t \approx 1$ for large load
- 2.7-2 $r/D = 1/4$, stress ratio = 1.14
- 2.7-3 $a/b = 2$, $\sigma_{\max} = 1.5\sigma_1$
- 2.7-4(a) $\nu = 1/3$ (b) $a/b = 1/\nu$, $\sigma_A = \sigma_B = -(1 + \nu)\sigma_O$
- 2.7-5(a) [Proof required] (b) $\sigma_{\max} = 159T/D^3$ (c) $T_{fp} = 0.0565\tau_Y D^3$
- 2.7-6 Cut away a central strip of width w
- 2.7-7 Residual $\sigma_B = \sigma_Y(1 - K_t)$ (compressive)
- 2.8-1(a) $p_O = 0.591\sqrt{PE/LR}$ (b) $p_O = 0.418\sqrt{PE/LR}$ (c) $p_O = 0.091\sqrt{PE/LR}$
- 2.8-2(a) $T = PR\phi$ (b) $p_O = 0.296(\phi/R)\sqrt{PE}$ (c) $\sigma = 298$ MPa
- 2.8-3 [Argument resembles that of Problem 1.7-8]
- 3.2-1(a) [Derivation required] (b) $\tau = \sigma_{tf}\sigma_{cf}(\sigma_{tf} + \sigma_{cf})$
- 3.2-2 Expand square in third quadrant of Fig. 3.3-1 to triple size
- 3.2-3(a) -240 MPa to 40 MPa (b) -120 MPa to 25 MPa
- 3.2-4 $T = 7.57$ kN·m
- 3.2-5 $r = 61.4$ mm
- 3.3-1 Results in first quadrant ($\sigma_x > \sigma_y > 0$; then $\sigma_y > \sigma_x > 0$):
- (a) 45° to x and z axes; then 45° to y and z axes
- (b) $\sigma_x = \sigma_1$, $\sigma_y = \sigma_2$, $0 = \sigma_3$; then $\sigma_y = \sigma_1$, $\sigma_x = \sigma_2$, $0 = \sigma_3$
- (c) $\sigma_x = \sigma_y$, then $\sigma_y = \sigma_y$
- 3.3-2(a) $\sigma_x^2 + 4\tau_{xy}^2 = \sigma_y^2$ (b) $\sigma_x^2 + 3\tau_{xy}^2 = \sigma_y^2$
- (c) $\sigma_1^2 - \sigma_1\sigma_3 + \sigma_3^2 = \sigma_y^2$ (d) $\sigma_x^2 - \sigma_x\sigma_y + \sigma_y^2 + 3\tau_{xy}^2 = \sigma_y^2$
- 3.3-3(a) 1,3,2 (b) 1,3,2 (c) 3, 1 and 2 tied (d) 3,2,1
- 3.3-4(a) $\sigma_y = 110$ MPa (b) $\sigma_y = 105.4$ MPa
- 3.3-5(a) -120 MPa to 50 MPa (b) -137.6 MPa to 67.6 MPa
- 3.3-6(a) $\sigma_y = 400$ MPa ($\tau_{\max}$ theory) or $\sigma_y = 346$ MPa (von M. theory)
- (b) $\sigma_y = 542$ MPa ($\tau_{\max}$ theory) or $\sigma_y = 542$ MPa (von M. theory)
- 3.3-7 $P = 62.8$ N ($\tau_{\max}$ theory) or $P = 69.2$ N (von M. theory)
- 3.3-8(a) $SF = 3.73$ (b) $SF = 4.11$
- 3.3-9(a) $t = 5.89$ mm (b) $t = 5.89$ mm
- 3.3-10(a) $r = 9.66$ mm (b) $r = 9.27$ mm
- 3.3-11 $r = [4(SF)\sqrt{M^2 + kT^2}/\pi\sigma_y]^{1/3}$: $k = 1$ in (a), $k = 0.75$ in (b)
- 3.5-1 $a = 5.24$ mm
- 3.5-2(a) $P = 1.51$ MN (b) $P = 4.52$ MN (c) $P = 1.54$ MN
- 3.5-3(a) $P = 173$ kN (b) $P = 59.2$ kN (c) $M = 1.29$ kN·m
- 3.5-4(a) $SF = 0.779$ (b) $SF = 0.728$ (c) $SF = 0.917$
- 3.5-5(a) $a = 28.1$ mm (b) $a = 24.7$ mm (c) $a = 20.3$ mm
- 3.5-6(a) $P = 12.4$ kN (b) $P = 31.9$ kN (c) $P = 17.9$ kN
- 3.5-7 First quadrant of ellipse with aspect ratio 0.75
- 3.5-8 [Rather lengthy expressions]
- 3.5-9(a) $T = 1.02$ MN·m (b) $T = 1.15$ MN·m (c) $T = 1.36$ MN·m
- 3.6-1 $N \approx 1000$ cycles
- 3.6-2 $2A = [(SF)(P_{\max} - P_{\min})/\sigma_{fs}] + [(P_{\max} + P_{\min})/\sigma_u]$
- 3.6-3(a) $SF = 1.08$ (b) $SF = 0.69$

- 3.6-4 Depth = 77.3 mm based on stress, 88.6 mm based on deflection
- 3.6-5(a) SF = 0.95 (b) SF = 0.52
- 3.6-6 SF = 4.74
- 3.6-7(a) About 26,000 cycles (b) About 600 repetitions
- 3.6-8(a) Yes (b) No (c) No (d) No (e) Yes
- 4.1-1 Energy expended = $w^2/4k$
- 4.1-2 $\theta = \arcsin(C/WL)$
- 4.1-3 $F_1 = k_1 a \theta$, $F_2 = 2k_2 a \theta$, where $\theta = C/[a^2(k_1 + 4k_2)]$
- 4.1-4 $\theta = \arcsin(W/2kL)$
- 4.1-5 $F_A = P/7$, $F_B = 2P/7$, $F_C = 4P/7$
- 4.1-6 $\theta = 4W/9ka$, $v = 13W/18k$
- 4.1-7 $U = (AEg^2/4L) + (P^2L/4AE)$
- 4.1-8(a) $u = (F/2\pi GL)\ln(R/r)$ (b) $\theta = (T/4\pi GL)(R^2 - r^2)/R^2r^2$
- 4.1-9 $T/\theta = 9GJ/20L$
- 4.2-1 $\theta = PL^2/2EI$
- 4.2-2 Change in length = Pvd/AE
- 4.2-3(a) $\Delta V = Fhr(1 - \nu)/2Et$ (b) $\Delta V = Fr^2(2 - \nu)/Et$
- 4.2-4 [Proof required]
- 4.2-5 [Explanation required]
- 4.2-6 [Proof required]
- 4.2-7 $\Delta V = Fh(1 - 2\nu)/E$
- 4.3-1 [Proof required]
- 4.3-2 [Proof required]
- 4.3-3(a,b,c) [Proof required]
- 4.4-1 $\theta = qL^3/6EI$, $v = 17qL^4/384EI$
- 4.5-1 $u_C = qL^2/2Eb$, $v_C = 2qL^3/Ebh^2$
- 4.5-2(a) $v_A = 14Fa^3/3EI$, $\theta_A = 2Fa^2/EI$
 (b) $v_C = 5Fa^3/6EI$, $\theta_C = 3Fa^2/2EI$
 (c) $\theta_{AC} = 23Fa^2/12EI$
- 4.5-3 $v_C = 5q_L L^4/768EI$, $\theta_C = 7q_L L^3/5760EI$
- 4.5-4(a) $u_A = 5QL^3/3EI$, $v_A = QL^3/EI$, $w_A = 0$
 (b) $u_A = 0$, $v_A = 0$, $w_A = (4FL^3/3EI) + (2FL^3/GK)$
- 4.5-5(a) $v_C = (qb^4/8EI) + (qa^3b/3EI) + (qab^3/2GK)$
 (b) $w_D = (qb^3c/6EI) + (qab^2c/2GK)$
 (c) $\theta_{xC} = (qb^3/6EI) + (qab^2/2GK)$
- 4.5-6 $\alpha = \pi/8$ or $\alpha = 5\pi/8$
- 4.5-7(a) 4.127PL/AE (rightward) (b) 8.954PL/AE (downward)
 (c) 0.752PL/AE (rightward) (d) 12.504PL/AE (downward)
 (e) 5.590PL/AE (separation)
- 4.6-1 $\theta_C = 1.15PR^2/EI$ at 60.3° clockwise from line AC
- 4.6-2 Exact: $v_C = 0.0621PL^3/EI$
 Simple approximation: $v_C = 0.0519PL^3/EI$
 Better approximation: $v_C = 0.0644PL^3/EI$

- 4.6-3 $u_O = CRL/EI$, $v_O = CL(R + L/2)/EI$,
 $w_O = (CL/EI)(R + L/2) + \pi(CR^2/4EI) - (CR^2/GJ)(1 - \pi/4)$
- 4.6-4(a) $u_A = 3\pi QR^3/EI$, $v_A = 0$, $w_A = 0$
 (b) $u_A = 0$, $v_A = 0$, $w_A = (\pi FR^3/EI) + (3\pi FR^3/GJ)$
- 4.6-5 Spring constant = $2EI/\pi R^3$
- 4.6-6 $u_A = 0$, $v_A = 0$, $w_A = \pi PR^3/GK$
 $\theta_{xA} = -2PR^2/GK$, $\theta_{yA} = 0$, $\theta_{zA} = 0$
- 4.6-7(a) $u_B = 2qR^4/3EI$ (b) $v_C = -0.226qR^4/EI$
- 4.6-8(a) $u_A = \pi^2 qR^4/EI$, $v_A = -3\pi qR^4/2EI$, $w_A = 0$
 (b) $u_A = 9\pi R^4/2EI$, $v_A = \pi^2 qR^4/EI$, $w_A = 0$
 (c) $u_A = 0$, $v_A = 0$, $w_A = 2\pi^2 qR^4/GJ$
- 4.6-9(a) $\theta_{xA} = 0$, $\theta_{yA} = 0$, $\theta_{zA} = 0$
 (b) $\theta_{xA} = 0$, $\theta_{yA} = 0$, $\theta_{zA} = 4\pi qR^3/EI$
 (c) $\theta_{xA} = \pi qR^3(3/GJ + 1/EI)$, $\theta_{yA} = 0$, $\theta_{zA} = 0$
- 4.6-10(a) $u_O = 0.163qR^4/EI$ (to right), $v_O = 0.215qR^4/EI$ (down)
 (b) $v_O = \pi qR^2/4EA$ (up)
- 4.6-11 $v = (FR^3/EI)[1 + \cos \phi + 0.5(\pi - \phi)\sin \phi]$
- 4.6-12(a) Use Eqs. 4.6-1; neglect effect of α (b) $w = 4PR^3n/Gc^4$
 (c) $\theta = 4nRC(2 + \nu)/Ec^4$ (d) $u = 2(2 + \nu)FH^3/3\pi Ec^4\alpha$
- 4.7-1 $a/b = 0.732$
- 4.7-2 Force = 0.85W
- 4.7-3 Separation = $Pb^3(4a + b)/[12EI(a + b)]$
- 4.7-4 $H_B = qa^3/[8b(a + b)]$
- 4.7-5 Reaction = $(5qa/4) - (6EIg/a^3)$
- 4.7-6 $v_A = 0.0709FL^3/EI$
- 4.7-7 $T = F/(2 + c)$, where $c = 6I/5AL^2$
- 4.7-8 $u_C = 20,900F/EL$
- 4.7-9 [Discussion required]
- 4.7-10 $M_C = (5Fa/16) + (qa^2/4)$
- 4.7-11 For $EI = GK$, $M_C = [Fa(a + 2b)/4 + qa^2(a + 3b)/6]/(a + b)$
- 4.7-12 $M_O = (FL/8) - (2\beta EI/L)$, $v_C = (FL^3/192EI) + (\epsilon L/4)$
- 4.7-13(a) $H \int_0^L y^2 ds = EI\alpha L\Delta T$ (b) $M = 0$ everywhere
- 4.7-14 [Discussion required]
- 4.7-15 $\theta = 0.149CR/EI$
- 4.7-16(a) $C = 0.307FR$ (b) $v = 0.0704FR^3/EI$
- 4.7-17(a) $M_C = 0.182PR$ (b) $M_A = 0.242PR$
 (c) $u_C = 0.0708PR^3/EI$ (d) $M_C = 0.151PR$
 (e) $u_B = -0.722PR^3/EI$ (f) $v_C = -0.0260M_C R^2/EI$

- 4.7-18(a) $v_C = 0.149PR^3/EI$ (b) $\Delta_{BD} = -0.137PR^3/EI$
 (c) $M_C = 0$, $w_C = (\pi R^2 M_O/4)(1/GK + 1/EI)$ (d) $T_C = M_O/\pi$
 (e) $M_C = 2PR/\pi[1 + (GK/EI)]$ (f) $M_B = 0.429 qR^2$
 (g) $v_C = -\rho R^5 \omega^2/6EI$
- 4.7-19 $M_A = (2T_O a/b)[(bEI + aGK)/(bEI + 2aGK)]$
- 4.7-20 $M_C = qL^2/8$, $Q = 5qL/8$
- 4.8-1 $w = 4PR^3 n/Gc^4$; lateral displacement increments cancel
- 4.10-1 $F_1 = -0.667P$, $F_2 = 0.0833P$, $F_3 = 0.750P$
- 4.10-2 $P = kL$; $d^2\Pi/d\theta^2 < 0$ if $P > kL$
- 4.10-3 $d^2\Pi/d\theta^2 > 0$ if $h < 2R$
- 4.10-4 $u_D = 2.083L\alpha\Delta T$, $v_D = 0$. Bar stresses are zero
- 4.11-1 $v_L = -qL^4/8EI$, $M_O = -qL^2/2$
- 4.11-2(a) $v_L = -FL^3/4EI$, $M_O = -FL/2$ (b) $v_L = -FL^3/3EI$, $M_O = -FL$
 (c) $v_L = -FL^3/12EI$, $M_O = 0$ (d) $v_L = -0.328FL^3/EI$, $M_O = -0.813FL$
- 4.11-3 $\theta_L = M_L L/3EI$
- 4.11-4(a) $v_C = -FL^3/64EI$, $M_C = FL/8$
 (b) $v_C = -qL^4/96EI$, $M_C = qL^2/12$
 (c) $v_C = -q_L L^4/192EI$, $M_C = q_L L^2/24$
- 4.11-5 $v_L = -0.308F/k$
- 4.11-6(a) $u = qLx/2EA$, $\sigma = qL/2A$
 (b) $u = (q/EA)(Lx - x^2/2)$, $\sigma = q(L - x)/A$
- 4.11-7(a) $v = ax^2(L - x)$ (b) $v_W = -0.00549WL^3/EI$
 (b) Stable if $EI > \gamma bL^4/420$
- 4.11-8 $P = 0.7222EA_0 u_L/L$ (0.12% high)
- 4.12-1 $v_O = 0.142FR^3/EI$, $M_O = 4FR/3\pi$
- 4.12-2 $v_O = \rho R^5 \omega^2/18EI$, $M_O = \rho R^3 \omega^2/9$
- 4.12-3 First term: v_L errs by -4.5%; M_O errs by -41%
- 4.12-4(a) First term: $v_L/2$ errs by -1.45%; $M_L/2$ errs by -18.9%
 (b) First term: $v_L/2$ errs by +0.38%; $M_L/2$ errs by +3.2%
 (c) First term: $v_L/2$ errs by +0.38%; $M_L/2$ errs by +3.2%
- 4.12-5 First term: u_L errs by +3.2%; σ_O errs by -18.9%
- 4.13-1 All results are exact
- 5.2-1 Deformation due to weight displaces equal weight of water.
- 5.2-2 $k = \rho g/l$
- 5.2-3 [Proof required]
- 5.2-4 $w = w_O e^{-\beta x}$, where $\beta^2 = k/T$
- 5.2-5(a) $\theta = -(T_O \lambda/k)e^{-\lambda x}$, $T = T_O e^{-\lambda x}$, where $\lambda^2 = k/GJ$
 (b) Replace T , θ , G , J by P , u , E , A respectively. $\lambda^2 = k/EA$
- 5.3-1(a,b) [Proof required]
- 5.3-2(a) $P_O = kw_O/\beta + k\theta_O/2\beta^2$, $M_O = kw_O/2\beta^2 + k\theta_O/2\beta^3$

- (b) $w = w_0 A_{\beta x} + (\theta_0/\beta) B_{\beta x}$, $\theta = -2\beta w_0 B_{\beta x} + \theta_0 C_{\beta x}$, etc.
- 5.3-3 Force = $0.2169P_0 = 10.8$ kN
- 5.3-4(a) $w_{\max}/w_{\min} = -0.2079$, $M_{\max}/M_{\min} = -23.1$
 (b) $F_+ = 0.6448\beta M_0$, $F_- = -0.6726\beta M_0$
- 5.3-5 $P_0 = 45.2$ kN. Upward w : initial uniform pressure decreases.
- 5.3-6 $\sigma = 190$ MPa, at $x = 99.9$ mm from loaded end
- 5.3-7 $w = (2\beta^2 M_0/k) B_{\beta x}$, $\theta = (2\beta^3 M_0/k) C_{\beta x}$, etc.
- 5.3-8(a) $a = 1/2\beta$
 (b) $M_{\max} = Fa$, at $x = 0$. $M_{\min} = -0.1040F/\beta$, at $\beta x = \pi/2$
- 5.3-9(a) $a = 1/\beta = 490$ mm (b) $F = 488$ N
- 5.3-10 $Q = 168$ kN
- 5.3-11 $w_B = P[(3EI/L^3) + (k/2\beta)]^{-1}$
- 5.3-12 Depth = 104.5 mm
- 5.3-13 $M_{\text{cusp}} = 2.71$ kN·m
- 5.3-14 $M_A = M_B = (\beta a - 1)P/4\beta$. AB is simply supported for $\beta a = 1$.
- 5.4-1 [Proof required]
 5.4-2 [Proof required]
 5.4-3 [Proof required]
 5.4-4 [Proof required]
- 5.4-5 $w_{\max} = P/K + Ps^3/192EI$, $M_{\max} = Ps/8$
- 5.4-6 [Argument required]
- 5.4-7 $\theta = M_0 h/[4EI(1 + \beta h)]$
- 5.4-8 Force load: factors 0.595 , 0.841
 Moment load: factors 0.707 , 1.000
- 5.4-9 $\sigma \approx 176$ MPa
- 5.5-1 $M = 13,860$ N·mm, $\Delta\sigma = 141$ MPa
- 5.5-2 Depth = 139 mm
- 5.5-3 Maximum $\sigma_{\text{long}} = 61.6$ MPa, maximum $\sigma_{\text{cross}} = 45.1$ MPa
- 5.5-4 Separation = $\pi/2\beta$
- 5.5-5 $k = 0.264(P^4/EI\alpha^4)^{1/3}$
- 5.5-6(a) $s \approx 0.179/\beta$ (b) $s = 0.0257/\beta$
- 5.5-7 Separation = $1.86/\beta$
- 5.5-8(a) $w_{\max} = 16.4$ mm, 200 mm left of load $2P$
 (b) $M_{\max} = 4.38$ kN·m, beneath load $2P$
- 5.5-9 $x = 5\pi/4\beta$, $P_0 = 71.9\beta M_0$
- 5.5-10 $a = 1280$ mm, $M = -4.73$ kN·m
- 5.5-11(a) $w_{\max} = 4.07$ mm, $\sigma_{\max} = 97.3$ MPa
 (b) $w_{\max} = 6.45$ mm (at center load),
 $\sigma_{\max} = 70.0$ MPa (at side load)
- 5.5-12(a) $\sigma = 0.0146P$ (b) $\sigma = 0.0226P$ (c) $\sigma = 0.0110P$
- 5.5-13(a) $w = 0.297$ mm, $M = 4.20$ kN·m
 (b) $w = 0.228$ mm, $M = 0.622$ kN·m
 (c) $w = 0.170$ mm
 (d) $w = 0.223$ mm, $M = 4.00$ kN·m at load P
- 5.5-14 [Derivation required]

- 5.5-15 New couple is $0.586M_0$ (acting on portion to right)
- 5.5-16 Bending moment = $0.268F/\beta$
- 5.6-1(a) $w = q(1 - D_{\beta x})/k$ (b) $M = (q/2\beta^2)B_{\beta x}$
 (c) $w = q(1 - A_{\beta x})/k$ (d) $M_{\max} = 0.104q/\beta^2$, $M_{\min} = -q/2\beta^2$
- 5.6-2 [Sketches required]
- 5.6-3 [Proof required]
- 5.6-4(a) $M_{\text{center}} = 8840q$ (b) $M_{\text{ends}} = -582q$ (c) $M_{\pi/4} = 27,847q$
 (d) $M_{\max} = 27,850q$ (e) $M_{\min} = -29,149q$
- 5.6-5 $M_{\text{center}} = 126,200q$, $M_{\text{supports}} = -215,000q$
- 5.6-6 $M_{\max} = 0.0806(q_2 - q_1)/\beta^2$
- 5.6-7 $w = (q/2k)(2 + B_{\beta l}C_{\beta a} - C_{\beta l}D_{\beta a} - D_{\beta b})$
 $M = (q/4\beta^2)(B_{\beta b} + B_{\beta a}C_{\beta l} - B_{\beta l}A_{\beta a})$
- 5.7-1 $w = -EI(d^2w/dx^2) = 0$ @ $x = 0$, $d^2w/dx^2 = d^3w/dx^3 = 0$ @ $x = L$
- 5.7-2 [Proof required]
- 5.7-3 $p = (2P/bL)(2 - 3a/L) + (6P/bL^2)(2a/L - 1)x$, $w = p/k_0$
- 5.7-4(a) $L/3 < a < 2L/3$ (b) $w_{\max} = 4P/3k_0bL$, $w_{\min} = P/k_0bL$
- 5.7-5 $P = 3kLw_0/8$
- 5.7-6(a) $M = PL/8$ (b) $M_{\min} = -4PL/27$, at $x = L/3$
- 5.7-7 Energy expended = $3W^2/2k_0bL$
- 5.7-8(a) $w_{\max} = (P/\pi R^2k_0)(1 + 4a/R)$ (b) $a = R/4$
- 5.7-9(a) $p = (F/4ab)(1 + 3x_0x/a^2 + 3y_0y/b^2)$
 (b) Central "diamond;" edge in 1st quadrant: $3x_0/a + 3y_0/b = 1$
- 6.1-1 $\tau = Gy(d\theta/d\phi)/(r_n - y)$
- 6.2-1 [Derivation required]
- 6.2-2 [Derivation required]
- 6.2-3 [Derivation required]
- 6.2-4 [Derivation required]
- 6.2-5(a) A , exact; $\int dA/r$, -0.1%; R , 0.5%; r_n , 0.1%; e , -0.7%
 (b) A , exact; $\int dA/r$, -0.02%; R , $\approx$ exact; r_n , 0.03%; e , -0.2%
- 6.2-6 $e = c^2/4R$; [proof required]
- 6.2-7 For $a/b = 1.2, 1.6, 3.0$, and 8.0 :
 (a) 0.0102, 0.0264, 0.0617, 0.1129
 (b) 0.0104, 0.0278, 0.0667, 0.1167
 (c) 1.056, 1.166, 1.469, 2.276
 (d) 0.922, 0.822, 0.656, 0.503
- 6.2-8 $a/b \approx 2.65$
- 6.2-9 $b/a = 0.194$
- 6.3-1 At $r = b$, $\sigma_\phi = 101$ MPa. At $r = a$, $\sigma_\phi = -224$ MPa
- 6.3-2 At $r = b$, $\sigma_\phi = 315$ MPa. At $r = a$, $\sigma_\phi = -77.5$ MPa
- 6.3-3(a) 13.3% low (b) 46.5% low
- 6.3-4 $P = 62.5$ kN
- 6.3-5 $F = 6.33$ kN
- 6.3-6 $t_i = 116$ mm
- 6.3-7 $s = 1.91$ mm
- 6.3-8 Stress probably 283 MPa at most

- 6.4-1(a,b,c,d) [Derivation required]
- 6.4-2(a) Radial force (not radial stress) largest at $r_1 = r_n$
 (b) [Proof required] (c) $r_1 = b \exp(1 - b/r_n)$
- 6.4-3 $\sigma_r = 14.7 \text{ MPa}$
- 6.4-4 $\tau_1 = 28.6 \text{ MPa}$, in straight parts but not much beyond
- 6.4-5(a) $\sigma_r \approx 21 \text{ MPa}$ (b) $\sigma_r \approx -101 \text{ MPa}$
 (c) $\sigma_r = 3M/2Rht$ (d) 2.8% high
- 6.4-6 $\sigma_r = -0.00937P/t$
- 6.4-7(a) $\sigma_r = 73.1 \text{ MPa}$ (b) $\sigma_r = 39.4 \text{ MPa}$
- 6.4-8(a) $\sigma_r = 34.4 \text{ MPa}$ (b) $t = 18.9 \text{ mm}$
- 6.4-9 [Sketches required]
- 6.4-10 $\sigma_y = -2P^2x^2/Efb^2th^3$
- 6.5-1 [Sketches required]
- 6.5-2 $\sigma_\phi = -107 \text{ MPa}$ (inner), $\sigma_\phi = 169 \text{ MPa}$ (outer), $\tau_{\max} = 218 \text{ MPa}$
- 6.5-3(a) $\sigma_\phi = 13.0(10^{-6})M$, $\sigma_z = \pm 19.7(10^{-6})M$ } MPa if M is N·mm
 (b) $\sigma_r = 4.5(10^{-6})M$ (c) $\tau_{\max} = 16.3(10^{-6})M$
 (d) Reduce flange width and increase its thickness
- 6.5-4(a) $\sigma_\phi = 166 \text{ MPa}$ (inside), $\sigma_\phi = -244 \text{ MPa}$ (outside)
 (b) $\sigma_z = \pm 234 \text{ MPa}$ (c) Deforming ÷ nondeforming = 0.83
 (d) $\sigma_r = 65.6 \text{ MPa}$ (e) $\tau_{\max} = 200 \text{ MPa}$
- 6.5-5(a) $\sigma_\phi = 338 \text{ MPa}$ (inside), $\sigma_\phi = -110 \text{ MPa}$ (outside)
 (b) $\sigma_z = \pm 172 \text{ MPa}$ (c) Deforming ÷ nondeforming = 0.85
 (d) $\sigma_r = 36 \text{ MPa}$ (e) $\tau_{\max} = 131 \text{ MPa}$
- 6.6-1 [Sketches required]
- 6.6-2(a) $\sigma_r \approx 0.68 \text{ MPa}$ (b) $\tau_{\max} \approx 4.5 \text{ MPa}$
- 6.7-1 [Derivation required]
- 6.7-2 $v_o = (FR/EA)(0.785 + 5.318 + 0.429 + 2.180)$. Ratio = 1.53
- 6.7-3 With $q = pta/R$, $v_A = (\pi q R^2/2A)[3(R/e - 1)/E + k/G] = 1575p/E$
- 6.7-4(a) [Derivation required]
 (b) $v_r = \frac{PR}{EA} \left[\frac{4}{\pi} - \frac{\pi}{4} - \frac{2e}{\pi R} + \left(\frac{\pi}{4} - \frac{2}{\pi} \right) \frac{R}{e} + \frac{\pi k(1+\nu)}{2} \right]$
 (c) Ratio = 2.89
- 7.1-1 $A_s = A_p(E_p/E_s)x/(L - x)$
- 7.1-2(a) $\sigma_x = q(L - x)/A$ (b) $u = (q/EA)[Lx - (x^2/2)]$
 (c) $\sigma_x = q(L - x)/A$ (d) $d(A\sigma_x)/dx + q = 0$; $q = q(x)$
- 7.2-1 $\sigma_x = 0$, $\sigma_y = 2Ea_1x$, $\tau_{xy} = 0$ (pure bending)
- 7.2-2 $\epsilon_{x'} = \epsilon_x \cos^2 \theta + \epsilon_y \sin^2 \theta + \gamma_{xy} \sin \theta \cos \theta$
 $\gamma_{x'y'} = 2(\epsilon_y - \epsilon_x) \sin \theta \cos \theta + \gamma_{xy}(\cos^2 \theta - \sin^2 \theta)$
- 7.2-3 $\sigma_x = (qy/2I)(x - L)^2/(1 - \nu^2)$. M correct if $\nu = 0$
- 7.3-1(a) $\partial N_x/\partial x + \partial N_{xy}/\partial y + B_x t = 0$, etc. (b) [Derivation required]
- 7.3-2(a) Equilibrium not satisfied (b) Valid
 (c) Equilibrium not satisfied (d) Valid
- 7.3-3 $\tau_{xy} = (3P/4c)[1 - (y^2/c^2)]$

- 7.3-4 $\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} + \frac{1+\nu}{1-\nu} \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 v}{\partial x \partial y} \right) = 0$, etc.
- 7.3-5 [Sketches required]
- 7.3-6 $u = 0$ on x and y axes; top edge and curved edge free;
 $\sigma_x = My/I$ on right edge. Set $v = 0$ at some point.
- 7.4-1(a,b) [Derivation required]
- 7.4-2 Equilibrium satisfied, but not compatibility across $x = 1$
- 7.4-3 [Sketches required]
- 7.4-4 $F = (a_1 y^3 + a_2 x^3)/6 + c_1 xy + c_2 y + c_3 x + c_4$, $\tau_{xy} = -c_1$
- 7.4-5 $g(y) = 5 - 43.7y$, $f = -16.3x - 6$, compatibility satisfied
- 7.4-6 [Argument required]
- 7.5-1(a,b,c,d) [Proof required]
- 7.5-2 [Proof required]
- 7.5-3 [Proof required]
- 7.5-4 Add $qx^2/2$ to right hand side of Eq. 7.5-7
- 7.5-5 $\tau_{xy} = (qx/2I)(c^2 - y^2)$, $\sigma_y = -(c^2 y/2 - y^3/6 + c^3/3)q/I$
 Compatibility not satisfied
- 7.5-6 [Discussion required]
- 7.6-1 $u = -\nu \rho g x(y - L)/E + c_1 y + c_2$
 $v = \rho g(y^2 - 2Ly + \nu x^2)/2E - c_1 x + c_3$ } $c_2 = c_3 = 0$
 may set $c_1 = 0$
- 7.6-2 $\sigma_x = \rho g x$, $u = (\rho g/2E)[x^2 - L^2 + \nu(y^2 + z^2)]$,
 $v = -\nu \rho g xy/E$, $w = -\nu \rho g xz/E$
- 7.6-3 $u = [a_1 xy - a_2(y^2 + \nu x^2)/2]/E + a_3 y/G$
 $v = [a_2 xy - a_1(x^2 + \nu y^2)/2]/E$
- 7.6-4 $u = P[3(x^2 - L^2)y + \nu y^3]/4Ec^3 + Py(3c^2 - y^2)/4Gc^3$
 $v = P[L^2(3x - 2L) - x(x^2 - 3\nu y^2)]/4Ec^3$
- 7.7-1 [Proof required]
- 7.7-2 $\sigma_A = 42 \text{ kPa}$, $\sigma_B = -33.8 \text{ kPa}$
- 7.7-3(a,b,c) [Proof required]
- 7.7-4 Uniaxial stress $\sigma_y = 4a_1$ (in direction $\theta = \pi/2$)
- 7.7-5 Cylinder, internal pressure (Eqs. 8.2-2)
- 7.7-6(a) Stresses due to torque $T = 2\pi C$
 (b) $F = A \sin 2\theta + B \cos 2\theta + C\theta + D$ ($A, B, C, D = \text{constants}$)
- 7.7-7(a) $\sigma_r = \sigma_{\theta a} \ln(r/a)$ (b) [Proof required]
 (c) [Derivation required] (d) $|M| = (2a^3 \sigma_{za}/9) \sin(\alpha/2)$
 (e) Green wood is stronger in tension than in compression
- 7.8-1 $g = a_5 r^4 + a_6 r^2 + a_7/r^2 + a_8$
- 7.8-2(a) $F_y = 0$ (b) $F_x = 9.9495 \sigma_0 b$
- 7.8-3(a) SCF = 2 (b) SCF = 4
- 7.8-4(a) $\theta = \pm \pi/4$ or $\pm 3\pi/4$ ($\sigma_\theta > 0$); $\theta = 0$ or π ($\sigma_\theta < 0$)
 (b) For example: $\sigma_x = \sigma_0$, $\sigma_y = 3\sigma_0$
- 7.8-5 $\sigma_\theta = 137 \text{ MPa (max)}$, $\sigma_\theta = -97.1 \text{ MPa (min; } 90^\circ \text{ away)}$
- 7.9-1(a) [Proof required] (b) [Proof required]
 (c) Top surface: $\sigma_r = 49.29P/r$, $Mc/I = 48.99P/r$
 Midline: $\sigma_{r\theta} = 0$, $VQ/It = 4.32P/r$
- 7.9-2 $\sigma_1 = -\sigma_3 = \sqrt{2} P/\pi a$, $\sigma_2 = 0$

- 7.9-3 $\sigma_x = -\frac{2q}{\pi} \left[\arctan \frac{a}{b} - \frac{ab}{a^2 + b^2} \right]$, $\sigma_y = -\frac{2q}{\pi} \left[\arctan \frac{a}{b} + \frac{ab}{a^2 + b^2} \right]$
- 7.9-4 $F = (M/\pi)(\theta + \sin \theta \cos \theta)$
- 7.9-5(a) [Proof required]
 (b) E.g. at $y = -c$: $\sigma_x = 2.68P/c$, $Mc/I = 3.00P/c$
- 7.9-6(a) $\sigma_r = (-P/r)[2.141 \cos \beta - 1.363 \sin \beta]$ (b) [Proof required]
- 7.9-7(a) $\sigma_r = (2P/\pi r) \cos(\beta + \lambda)$ (b) $\beta + \lambda = \pi/2$
 (c) Line OA collinear with force P
- 7.9-8 [Proof required]
- 7.10-1 For $T = T_0$: $a_1 = 0$, $a_2 = E\alpha T_0$.
 For $T = T_0 y/c$: $a_1 = E\alpha T_0/c$, $a_2 = 0$
- 7.10-2 $\sigma_x = -E\alpha T_0 y/c = -E\alpha T$
- 7.10-3 $\sigma_x = \sigma_y = -E\alpha z/(1 - \nu)$, $\sigma_z = 0$
 $u = v = 0$, $w = (\alpha a z^2/2)[(1 + \nu)/(1 - \nu)]$
- 7.10-4 $\sigma_\theta = \sigma_z = \pm (E\alpha/2)(T_a - T_b)/(1 - \nu)$; + inside, - outside
- 7.10-5(a,b,c) [Proof required]
- 7.10-6 $\sigma_r = E\alpha k(r^2 - a^2)/4$, $\sigma_\theta = E\alpha k(3r^2 - a^2)/4$
- 7.10-7 $T = T_b + (T_a - T_b) \ln(r/b)/\ln(a/b)$
- 7.11-1(a,b) [Proof required]
- 7.11-2 $u = T(b^2 - a^2)yz/\pi G a^3 b^3$
- 7.12-1 [Proof required]
- 7.12-2 $\beta = (T/\pi a^3 b^3 G)(a^2 + b^2)/(1 - k^4)$, $\gamma_{\max} = (2T/\pi a b^2)/(1 - k^4)$
- 7.12-3(a) $k = G\beta/2h$, $\beta = 15\sqrt{3} T/Gh^4$, $\gamma_{\max} = 15\sqrt{3} T/2h^3$
 (b) Fails; $\nabla^2 \phi = -2G\beta$ not satisfied
- 7.12-4(a) [Proof required] (b) [Proof required]
 (c) Stress ratio = $4a(b - 2a)/(4a^2 - b^2)$. SCF $\rightarrow 2$ as $b \rightarrow 0$
- 8.2-1 [Proof required]
- 8.2-2 At $r = b$: $\sigma_\theta = p_i$ and $\sigma_\theta = 1.5 p_i$ respectively
- 8.2-3(a) $p_i = 87.2$ MPa (b) $p_i = 101$ MPa
- 8.2-4 $p_i = 139$ MPa
- 8.2-5 $F_i = 11/16$ of total; $W_i = 3/8$ of total
- 8.2-6(a) $a^2 = b^2(\sigma_{\max} + p_i)/(\sigma_{\max} - p_i)$
 (b) $a^2 = b^2\sigma_y/(\sigma_y - 2p_i)$
 (c) $a^2 = b^2(\sigma_y^2 + p_i\sqrt{4\sigma_y^2 - 3p_i^2})/(\sigma_y^2 - 3p_i^2)$
 (d) $a = 1.291b$, $a = 1.414b$, $a = 1.353b$
- 8.2-7(a) $p_o = (p_i/2)(1 + b^2/a^2)$, $\sigma_\theta = -(p_i/2)(1 - b^2/a^2)$
 (b) $\gamma_{\max} = p_i/2$ at $r = b$
- 8.2-8 $1 < (p_i/p_o) < (3a^2/b^2 + 1)/(a^2/b^2 + 3)$
- 8.2-9(a) $2(\sigma_t - \sigma_r) - r(d\sigma_r/dr) = 0$
 (b) $(d^2u/dr^2) + (2/r)(du/dr) - 2u/r^2 = 0$
 (c) $u = Ar + B/r^2$; A and B are integration constants
- 8.3-1 [Sketches required]
- 8.3-2 $p_i = 6 + 1.52p_c$

8.3-3 [Several equations constitute the required answers]

8.3-4(a) $\Delta = (2a^2bp_C/E)/(a^2 - b^2)$

(b) $\Delta L = (2va^2Lp_C/E)/(a^2 - b^2)$

8.3-5 $p_C = 9p_i/16$

8.3-6 $p_i = 1.215\bar{\sigma}_\theta$, $p_C = 0.1947\bar{\sigma}_\theta$

8.3-7 $2\Delta = 0.0907$ mm

8.3-8 [Proof required]

8.3-9 $\sigma_\theta = -144kp_O/[25 + 39k - 15(1 - k)\nu]$

8.3-10(a) $p_C = 9.375$ MPa (b) $p_i = 34.8$ MPa

8.3-11 [Proof required]

8.3-12 [Proof required]

8.3-13 [Proof required]

8.3-14(a) $\sigma_r = -87.3$ MPa (b) $\sigma_\theta = 216$ MPa (c) $\sigma_\theta = 216$ MPa

8.3-15(a) $a = 9.09$ mm, $c = 6.74$ mm

(b) $\sigma_e = 347$ MPa at $r = b$, $\sigma_e = 363$ MPa at $r = c$

(c) $\sigma_e = 351$ MPa at $r = b$, $\sigma_e = 349$ MPa at $r = c$

8.3-16 Factor = 1.25 for both

8.3-17 $\tau_{\max} = 3\sqrt{3} p_i b/2a$ at $r = c$, $p_i = 0.1925\sigma_Y(a/b)$

8.4-1(a) [Proof required] (b) $\rho_s/\rho = 3E_s/E$ (s for spokes)

8.4-2 [Proof required]

8.4-3 [Proof required]

8.4-4 $a/b = 1.08$

8.4-5 [Proof required]

8.4-6 $\sigma_r = -89.3$ MPa, $\sigma_\theta = 161$ MPa

8.4-7 $T = 57.4(10^6)$ N·mm, $\tau_{\text{net}} = 85.8$ MPa

8.4-8(a) $\omega = 19,900$ rpm (b) $\omega = 13,850$ rpm

8.4-9 Yes (barely)

8.4-10(a) $\sigma_O = (\rho\omega^2/6a)(c^3 - a^3)$ (b) $\sigma_\theta = 91.7$ MPa

8.4-11(a) $\omega_O^2 = 8p_C/[(3 + \nu)\rho a^2(1 - b^2/a^2)]$ (b) $\omega = \omega_O/\sqrt{3}$

(c) $P = 4\pi\mu p_C b^2 h\omega_O/(3\sqrt{3})$

8.4-12(a) $p_C = 42.15$ MPa (b) $\omega = 6450$ rpm

8.4-13 [Explanations required]

8.5-1 $h_O = 61.1$ mm

8.5-2(a) $h_O = 25.7$ mm, $h_{O.2} = 25.0$ mm, $h_{O.4} = 23.0$ mm

(b) $h_O = 54.7$ mm, $h_{O.2} = 48.9$ mm, $h_{O.4} = 35.0$ mm

(c) $h_O = 84.1$ mm, $h_{O.2} = 71.7$ mm, $h_{O.4} = 44.4$ mm

(d) $h_O = 141.3$ mm, $h_{O.2} = 113.7$ mm, $h_{O.4} = 59.3$ mm

8.5-3(a) $k = \rho\omega^2 \left[\int_b^a hr^2 dr / \int_b^a \frac{h}{r} dr \right]$ where $h = h(r)$

(b) Error = 14.2%

8.6-1(a) $\sigma_r = \sigma_Y \ln(r/a)$, $\sigma_\theta = \sigma_Y + \sigma_r$

(b) $p_{fp} = \sigma_Y \ln(a/b)$

8.6-2(a) $a = 317.5$ mm

(b) $a = 210$ mm, $W_b/W_a = 0.344$

(c) $a = 184$ mm, $W_c/W_a = 0.226$

- 8.6-3 $p_{fp} = 352 \text{ MPa}$
- 8.6-4 Single: impossible. Compound: $a/b = 4.93$, wt. ratio = 5.92
- 8.6-5 $p_{fp} = 2\sigma_Y \ln(a/b)$
- 8.6-6 $p_{fp} = 605 \text{ MPa}$. Yielding before new p_i reaches p_{fp}
- 8.6-7 $p_{fp} = 316 \text{ MPa}$. Yielding when new p_i reaches p_{fp}
- 8.6-8 [Proof required]
- 8.6-9 $p_i = 0.9375\sigma_Y$, $c = 16.94 \text{ mm}$
- 8.6-10(a) $2(a/b)^2 \ln(a/b) / [(a/b)^2 - 1] = 1 + \beta$
 (b) $a/b = 1.548$ (c) $a/b = 2.22$
- 8.7-1 [Derivation required]
- 8.7-2(a) $\omega_{fp}^2 = 2\sigma_Y \ln(a/b) / [\rho(a^2 - b^2)]$ (b) Factor = 1.267
- 8.7-3(a) $\sigma_r = \sigma_Y(1 - r^2/a^2)$, $\sigma_\theta = \sigma_Y$ for all r
 (b) $\sigma_r = \sigma_Y(1 - 0.9524b/r - 0.0476r^2/b^2)$, $\sigma_\theta = \sigma_Y$ for all r
- 8.7-4(a) $\omega_{fp}^2 = 5.25\sigma_Y/\rho$ (b) Factor = 0.619 (c) No (SCF neglected)
- 8.7-5 [Proof required]
- 8.7-6 $a/b = 4.85$
- 9.2-1 $\beta = T/GJ$, $\tau_{max} = TR/J$, where $J = \pi(R^4 - b^4)/2$
- 9.3-1 Error = 303% (high)
- 9.3-2(a) $\tau = 3T/[2(1 + \pi)Rt^2]$, $\beta = \tau/Gt$
 (b) τ ratio = 25.2, β ratio = 252
- 9.3-3 $c/h = 3/7$, β ratio = 4, $T_{thicker} = 0.857T_{total}$
- 9.3-4 $GK = 63,280G \text{ N}\cdot\text{mm}^2$, $\tau_{max} = 237(10^{-6})T \text{ MPa}$
- 9.3-5(a) $\tau_{max} = 12T/a^2b$, $\beta = 12T/Ga^3b$
 (b) 100% error (high) for both
- 9.3-6 $\sigma = 3PL/2a^2t$, $\tau = (3P/4t)(1/a + 1/t)$
- 9.4-1(a) $\tau_{max} = 2T/\pi ab^2$ (b) $T/\beta = 3.101Ga^3b^3$
- 9.4-2(a) Ratio = 1.354 (b) Ratio = 0.678
 (c) Square: $T/\beta = 2.26Ga^4$ (table), $T/\beta = 2.40Ga^4$ (equation)
 Rectang.: $T/\beta = 1.12Ga^4$ (table), $T/\beta = 1.13Ga^4$ (equation)
- 9.4-3 Square: $\tau = 4.81T/A^{1.5}$, $\beta = 7.09T/GA^2$
 Circle: $\tau = 3.545T/A^{1.5}$, $\beta = 6.28T/GA^2$
- 9.4-4 Allowable $T = 144 \text{ kN}\cdot\text{mm}$, $\theta = 1.85^\circ$
- 9.5-1 [Derivation required]
- 9.5-2(a) τ ratio = $(1 + \eta^2)/(1 + \eta)$, β ratio = $2(1 + \eta^2)/(1 + \eta)^2$,
 where $\eta = b/a$
 (b) [Plot required] (c) [Proof required]
- 9.5-3 $R = 3000 \text{ mm}$ for open tube. May buckle
- 9.5-4 [Derivation required]
- 9.5-5(a) [Proof required] (b) τ ratio = 1.27, β ratio = 1.62
- 9.5-6(a) $T/\beta = 9.07GR^3t_o$; $\tau_{max} = 0.159T/R^2t_o$, outside at $\alpha = 0$
 (b) $T/\beta = 9.425GR^3t_o$; $\tau_{max} = 0.106T/R^2t_o$, outside at $\alpha = 0$
 (c) $\tau_{max} = 0.255T/Rt_o^2$, inside and very near $\alpha = \pm \pi$
- 9.5-7 $F = Ts/2\pi R^2$
- 9.5-8 $\theta = \frac{T}{4\pi Gt} \left(\frac{L}{r_i - r_o} \right)^3 \frac{L(2H+L)}{H^2(H+L)^2}$

- 9.6-1 [Derivation required]
- 9.6-2 $l_i/t_i \Gamma_i$ same in all cells i ; l_i = length of outer cell wall
- 9.6-3 Factor = 200
- 9.6-4 q decreases 4%, β increases less than 1%
- 9.6-5(a) $T/\beta = 10.83Ga^3t$, $q_{outer} = 0.0924T/a^2$, $q_{inner} = 0.0653T/a^2$
 (b) $T/\beta = 2\pi Gt(a^3 + b^3)$, $q_{outer} = Ta/[2\pi(a^3 + b^3)]$,
 $q_{inner} = Tb/[2\pi(a^3 + b^3)]$
- 9.6-6 $T = 48,700 \text{ N}\cdot\text{mm}$, $\beta = 3.53/G$ per mm
- 9.6-7 γ factor = β factor ≈ 1.78
- 9.7-1 Factor = 0.5; > 0.5 if noncircular (restraint of warping)
- 9.7-2(a) $\theta'''' - k^2\theta'' = -k^2T_q/GK$
 (b) Free: $\theta''' = 0$, $\theta'''' - k^2\theta'' = 0$
 Fixed: $\theta = 0$, $\theta' = 0$
 Simply supported: $\theta = 0$, $\theta'' = 0$ } $\theta' = \frac{d\theta}{dx}$, $\theta'' = \frac{d^2\theta}{dx^2}$, etc.
- 9.7-3(a,b) $\sigma_x = 16.1 \text{ MPa}$, $\theta = 9.02(10^{-3}) \text{ rad}$
- 9.7-4(a) $\sigma = \pm 91.6 \text{ MPa}$ (b) 2.88 mm left, 0.45 mm up
- 9.7-5 Midspan: $\sigma_x = 2130(10^{-6})P$
 Ends: $\tau_{zx} = 302(10^{-6})P$ in web, $\tau_{xy} = 261(10^{-6})P$ in flanges
- 9.8-1 [Plots required]
- 9.8-2 [Plots required]
- 9.9-1(a) $\omega = -1.261a^2$ at flange tips (b) $J_\omega = 3.363a^5t$
- 9.9-2(a) $\omega = \pm 8a^2/7$ at flange tips (b) $J_\omega = 1.905a^5t$
- 9.9-3(a) $\omega = \pm 2a^2$ at cut (b) $J_\omega = 3.70a^5t$
- 9.9-4(a) $\omega = (4R^2/\pi)\cos\alpha + R^2[\alpha - (\pi/2)]$ (b) $J_\omega = 0.0374R^5t$
- 9.9-5 $J_\omega = 7b^5t/24 + b^4ht/16$
- 9.10-1 $u_A = u_C = 0$, $u_B = u_D = (T/4Gb)(b/t_b - h/t_h)$
- 9.10-2 [Proof required]
- 9.10-3 $u = \pm 0.213 \text{ mm}$ on top, $u = \pm 0.0838 \text{ mm}$ on bottom
- 9.10-4 [Proof required]
- 9.11-1 $f = (\alpha^2/2) + 2(1 - \cos\alpha) - \pi\alpha$
- 9.11-2 q_{max} in flange = $E(d^2\beta/dx^2)(0.653a^3t)$
- 9.12-1 At support: $\sigma_x = 0.285P$ at slit, $\sigma_x = -0.166P$ on top
 At end: $\tau = 0.0284P$, $\theta_L = 0.000119P$
- 9.12-2 $T = 27,000 \text{ N}\cdot\text{mm}$
 At ends: $\sigma_x = 54.7 \text{ MPa}$ at flange tips, $q_{max}/t = 2.82 \text{ MPa}$
 At middle: $\tau_{sv} = 7.67 \text{ MPa}$
- 9.12-3 At support: $\sigma_x = 121 \text{ MPa}$, $q_{max}/t = 4.11 \text{ MPa}$
 At end: $\tau_{sv} = 14.5 \text{ MPa}$. θ reduction factor = 0.0645
- 9.12-4 $\sigma_x = 0.286P$
- 9.12-5 $L/a = 8$
- 9.12-6 Four constants: $\beta = 0$ at $x = 0$, $d\beta/dx = 0$ at $x = a + b$,
 β and $d\beta/dx$ must both match between parts at $x = a$
- 9.13-1(a) $\theta_L = -0.073B_L/Gt^4$, $\sigma_x = \pm 82.0(10^{-6})B_L/t^4$

- (b) $\theta_O = +0.075B_L/Gt^4$
- 9.13-2 $\tau/\sigma_x = 0.413$, $\tau_q/\sigma_x = 0.0413$
- 9.13-3 [Explanations required]
- 9.13-4 [Proof required]
- 9.14-1 $\sigma_O = 102 \text{ MPa}$, $\sigma_x = 121 \text{ MPa}$, $\tau_{SV} = 32.7 \text{ MPa}$
- 9.14-2 $T = 2470 \text{ N}\cdot\text{mm}$, $\tau_{SV} = 154 \text{ MPa}$
- 9.14-3(a) Factor = 3.60 (b) $\tau_{SV} = 0.0379T$, $\sigma_x = 0.127T$
- (c) $\Delta = -0.0244TL/G$ (d) $\Delta = 0.0332FL/G$
- 9.14-4(a) Factor = 3.11 (b) $\sigma_O = -113 \text{ MPa}$
- 9.15-1 [Proof required]
- 9.15-2 [Sketches required]
- 9.15-3(a,b) $T_{fp} = (\tau_{yt}^2/2)[a + b - (4t/3)]$ (c) $T_{fp} = 19.06\tau_y a^3$
- 9.15-4 $T = 31\pi\tau_y R^3/48$
- 9.15-5 $L/b = 8.3$
- 10.1-1(a) [Proof required] (b) $h/b = 1$
- 10.1-2 [Proof required]
- 10.1-3 $\sigma_x = N/A + B\omega/J_\omega + (\text{right hand side of Eq.10.1-5})$
- 10.2-1(a) [Proof required]
- (b) Factor: 0.707 for $h = b$, 0.503 for $h = 10b$
- 10.2-2 Diamond-shaped area with intercepts $y = \pm b/6$, $z = \pm h/6$
- 10.2-3 $\sigma_{xA} = 159 \text{ MPa}$, $\sigma_{xB} = -182 \text{ MPa}$
- 10.2-4 $\beta = -22.4^\circ$. Factor: 0.42 at A, 0.68 at B
- 10.2-5(a) $\lambda = -7.04^\circ$; $\sigma_{xA} = -105$, $\sigma_{xB} = -42.7$, $\sigma_{xC} = 148$ (all MPa)
- (b) $\beta = 76.0^\circ$
- 10.2-6(a) $\sigma_x = \pm 184 \text{ MPa}$, at flange tips
- (b) $\sigma_x = \pm 102 \text{ MPa}$, at web-flange intersection
- (c) $\sigma_x = 4.79 \text{ MPa}$, at right web-flange intersection
- 10.2-7 $M = 2.06(10^6) \text{ N}\cdot\text{mm}$ at $\beta = -60.9^\circ$
- 10.2-8 $R = 114 \text{ mm}$
- 10.2-9 $\sigma_x = -127 \text{ MPa}$ at upper flange tip
- $\sigma_x = 68.5 \text{ MPa}$ at lower corner
- 10.3-1(a) $27.7PL/b^3 \leq (\sigma_x)_{\text{tens}} \leq 32.0PL/b^3$
- (b) Deflection of tip = $18.48PL^3/Eb^4$ for all orientations
- 10.3-2 9.48° from horizontal
- 10.3-3 $P_y = -0.528P$, $P_z = -0.849P$
- 10.3-4(a,b) $\Delta = 0.340(10^6)q/E$, at 14.7° to vertical
- 10.3-5(a) $H = 244q$ (b) $\Delta = 280,000q/E$ (parallel to z axis)
- 10.3-6 $\Delta = 7.66 \text{ mm}$ at 31.4° above negative y axis
- 10.3-7 [Arguments required]
- 10.3-8 $\Delta = 0.115PL^3/EI$ at 34.6° below positive y axis
- 10.4-1 [Proof required]
- 10.4-2(a,b) [Proof required]
- 10.4-3 τ_{ave} is P/th at $x = 0$, $0.444P/th$ at $x = L/2$, $P/4th$ at $x = L$
- 10.4-4 $\alpha = 21.8^\circ$, $\tau = 0.00332V$

- 10.4-5(a) $q = V_z(1 - \cos \alpha)/\pi R$ (b) $q_{\max} = 2V_z/\pi R$ at $\alpha = \pi$
- 10.4-6(a) $z = -h/12$ (b) [Proof required]
 (c) $q_{\max} = 1.35V_y/h$ at $y = 3h/20$ (d) [Proof required]
- 10.4-7 $q_{\max} = .00990V$ at the centroid
- 10.4-8(a) $q = (3862 - 214.5z + 2.276z^2)10^{-6}V_y$
 (b) $q = (10,300 - 1.788y^2)10^{-6}V_y$
 (c) $q = 0$ at $z = \pm 24.24$ mm on flanges
 (d) About 19% low
- 10.5-1 Respective e_y 's, relative to $y = 0$: $-2R$, $-b/\sqrt{3}$, $-3\sqrt{2}b/4$
- 10.5-2 e_y is $2R[1 - (t/R)^2/3]$ left of origin
- 10.6-1 Rel. to vertical web (except as noted), $|e_y|$ distances are:
 (a) $3b^2(a^2 + c^2)/[2c^3 + 6b(a^2 + c^2)]$ (left)
 (b) $3b^2(c^2 - a^2)/[2c^3 + 6b(a^2 + c^2)]$ (left)
 (c) $3b^2/(6b + c)$ (right)
 (d) $0.155c$ (left)
 (e) $bc^3/(a^3 + c^3)$ (right of left flange)
 (f) $(b/2)(3b + 4c)/(3b + 2c)$ (left)
 (g) $(b/2)(3b + 4c)/(3b + 8c)$ (left)
 (h) $\sqrt{3}c/2$ (right of left vertex)
 (i) $\sqrt{3}h/4$ (left of left vertex)
 (j) $2R$ (left of centroid)
 (k) $2R(\sin \alpha - \alpha \cos \alpha)/(\alpha - \sin \alpha \cos \alpha)$ (left of cntr. of arc)
 (l) $0.510R$ (left)
- 10.6-2 e_y is $\pi R/2$ right of center of arc
- 10.6-3 Force = $3P/16$ on each weld, very localized near tip of beam
- 10.6-4(a) [Proof required]
 (b,c) [See answers for Problem 10.6-1, parts (e) and (h)]
- 10.6-5 Factor = 1.31
- 10.7-1 Relative to pole: $e_y = 44.7$ mm, $e_z = 164.9$ mm
- 10.7-2 [See answers already provided]
- 10.7-3 Relative to $x = y = 0$: $0.026a$ left, $1.021a$ below
- 10.8-1(a) $0.611R$ right of vertical web (b) $\beta = 2P/\pi^2 R^2 Gt$
- 11.2-1(a) $P_c = 56.8$ kN (b) $P_c = 24.0$ kN
- 11.2-2 Changes of slope at 0.7, 1.3, and 2.0 times $P/\sigma_y A$
- 11.2-3 $P_c = 1.25A\sigma_y$
- 11.2-4(a) $P_y = n\sigma_y A$ (b) $P_c = 4n\sigma_y A/\pi$
- 11.3-1(a) $f = 1.70$ (b) $f = 1.27$ (c) $f = 2.00$ (d) $f = 2.34$
- 11.3-2 $f = 1.137$
- 11.3-3 $M_{fp} = bh^2\sigma_y/12$
- 11.3-4 Linear for $0 < M < 3.60$ and for $4.52 < M < 5.11$ (kN·m)
- 11.3-5 [Proof required]
- 11.3-6(a) [Sketch required] (b) [Proof required]
 (c) $M = \sigma_y bc^2/3$ to yield again, $M_{fp} = \sigma_y bc^2$
- 11.3-7(a,b) [Sketch required]
- 11.3-8(a) $\rho = 7.143$ m (b) Residual $\rho = 13.8$ m
- 11.3-9(a,b) R of mandrel = 22.0 mm